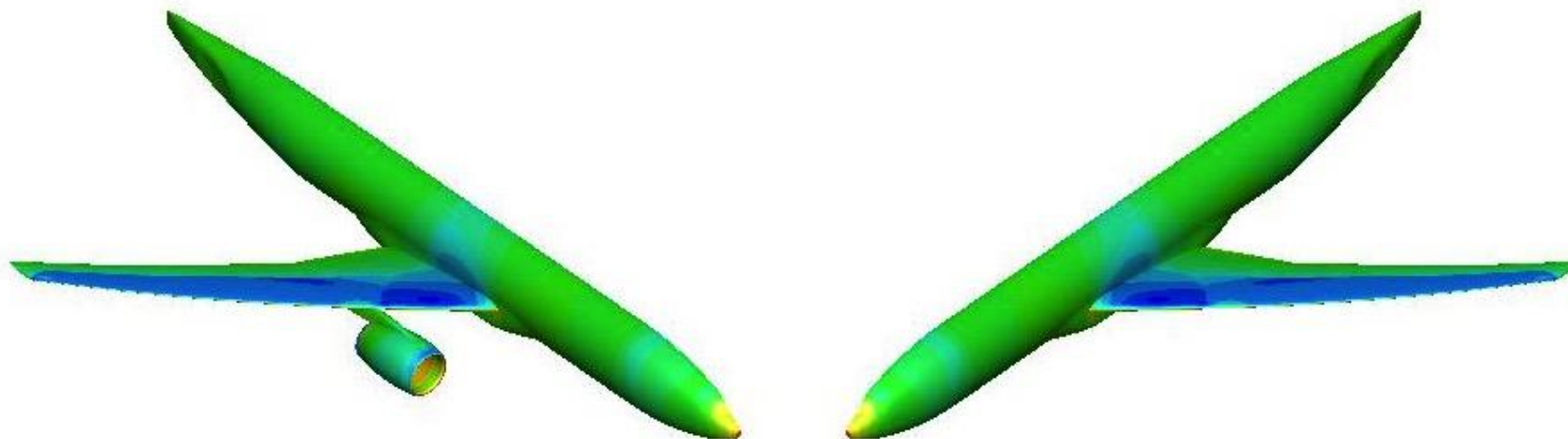


# DPW 6 Summary of Participant Data

## Case 1: Code Verification



Chris Roy (Virginia Tech)  
and the DPW Organizing Committee

## Outline:

- **Motivation for Case #1**
- **Problem description**
- **Participant data**
  - **Forces and moments**
  - Pressure and shear stress distributions
- **Effects to examine**
  - **Turbulence model**
  - **Grid type**
  - Order of accuracy of turbulence equations
- **Conclusions**

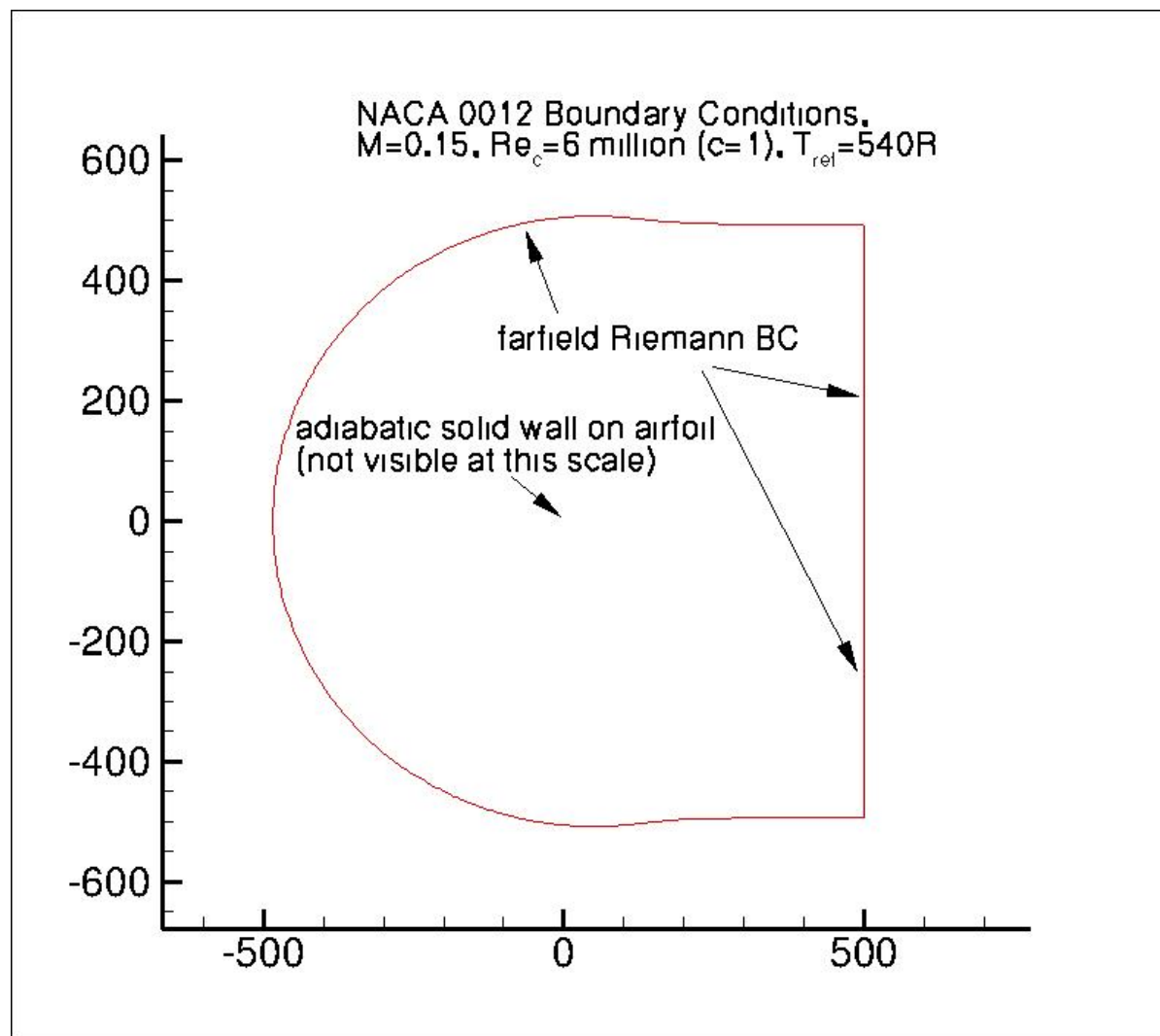
## Motivation for Case #1:

- **Past DPW Workshops had many instances of (ostensibly) the same turbulence model converging to different results with mesh refinement**
- **There should only be a single asymptotic answer for each model**
- **These problems could be attributed to a number of sources:**
  - Coding mistakes
  - Inconsistent algorithms or boundary conditions
  - Variations in the version of the turbulence model
  - Effects of grid type (structured, unstructured, etc.)
  - Ad hoc limiters on the turbulence equations (e.g., production)
  - Insufficient iterative convergence
- **Case #1 was designed to help identify contributions which suffered from these issues**

## Problem description: boundary conditions

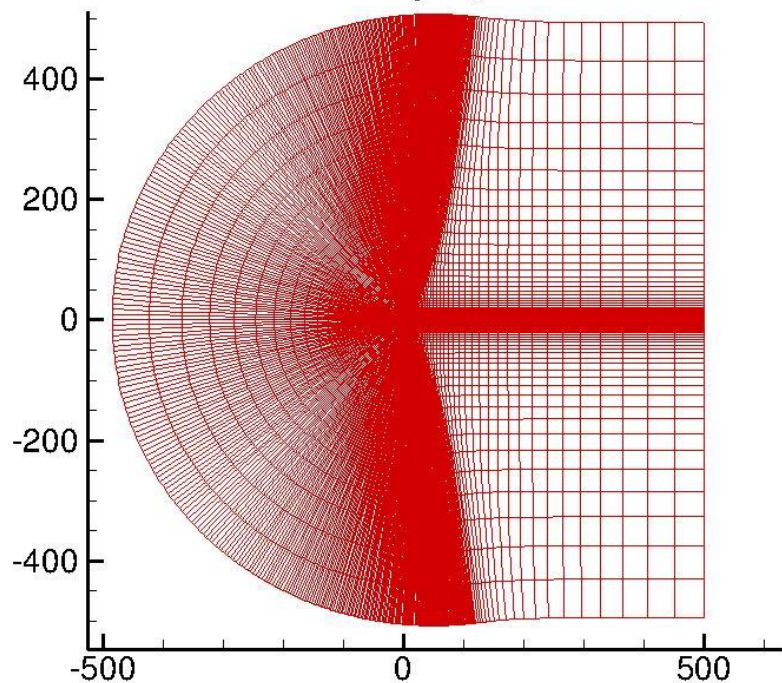
- NACA 0012 airfoil
- Mach 0.15
- $Re_c = 6M$
- Fixed Riemann BCs at ~500 chords
- Problem definition and grids supplied by the NASA Langley TurbModels web site:

[http://turbmodels.larc.nasa.gov/naca0012numerics\\_val.html](http://turbmodels.larc.nasa.gov/naca0012numerics_val.html)

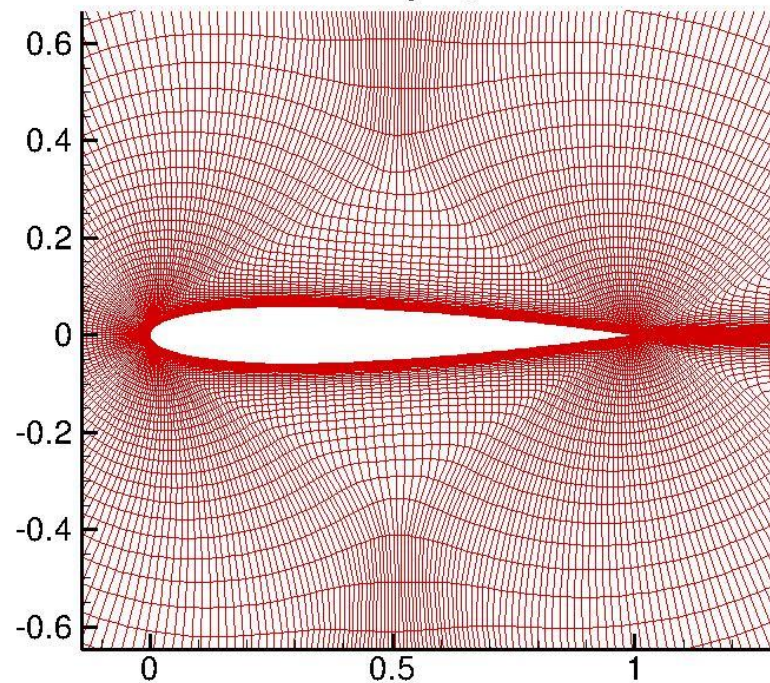


## Problem description: C-grids

449 x 129 grid, far view



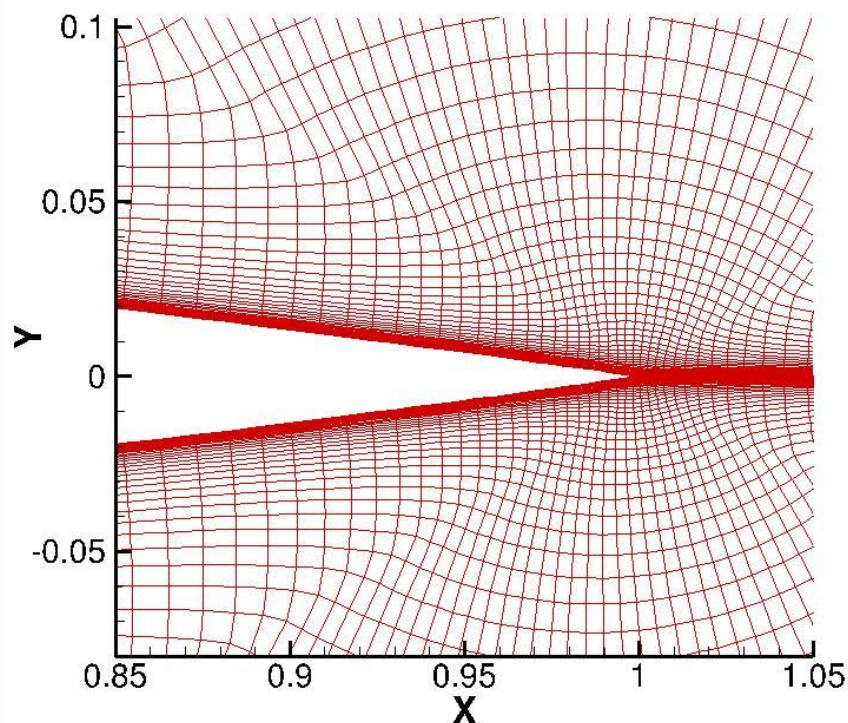
449 x 129 grid, near view





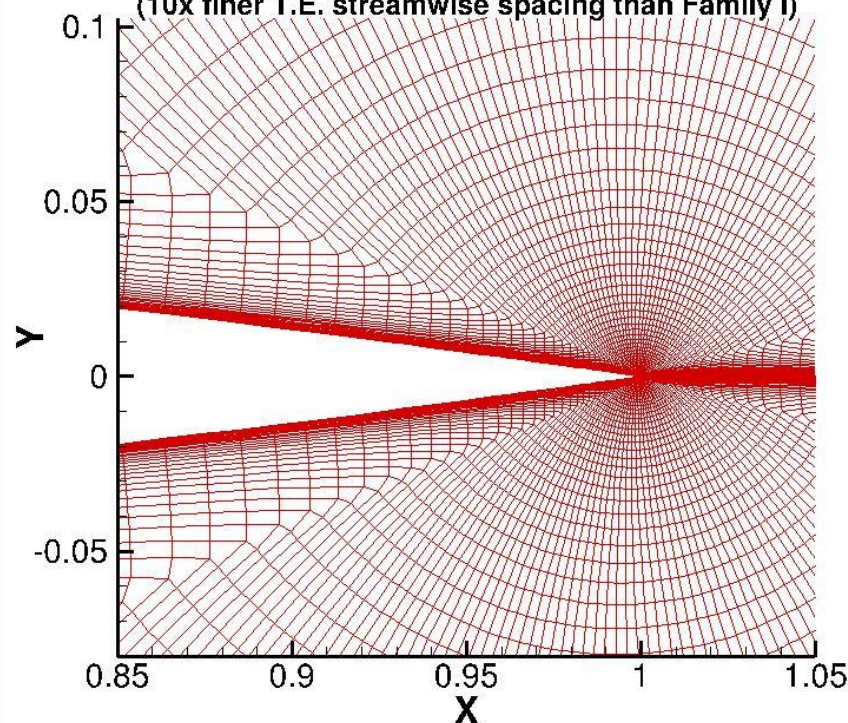
## Problem description: primary grids used

Family I, 449 x 129 grid (near view)



**Family I**

Family II, 449 x 129 grid (near view)  
(10x finer T.E. streamwise spacing than Family I)



**Family II (recommended)**

## Participant data summary for Case #1:

- **30 Data Total Data Submittals**
- **15 Teams/Organizations**
- **Turbulence Models:**
  - 22 SA (all types)
  - 4 SST
  - 1 k-kl, 1 k-e Lam, 1 EARSM, 1 LBM-VLES
- **Grid Types:**
  - 4      Structured Grid Family I (2 teams)
  - 21     Structured Grid Family II (11 teams) ← recommended grids
  - 1      O-Grid
  - 2      Cartesian (2 teams)
  - 1      Unstructured
  - 1      Adapted Unstructured

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Organisation	Name	Soln ID	Code	Turbulence Model	Case 1							
					Family	113x33	225x65	449x129	897x257	1793x513	3585x1025	7169x2049
Boeing, BCA Advanced Concepts, Long Beach CA	John Vassberg	A1	Overflow v2.2k	SA-RC	II	X	X	X	X	X		
		A2	Overflow v2.2k	SA-RC-QCR2000	II	X	X	X	X	X		
NASA Langley Research Center, FUN3D	Khaled S. Abdol-Hamid	B2	FUN3D 12.8	k-kL-MEAH2015	II	X	X	X	X	X	X	
CARD C	Jiangtao Chen	D1	Mflow	SA	II	X	X	X	X	X		
		D3	Mflow	SST	II	X	X	X	X	X		
	Yan Sun	T1	TRIP	SA	I	X	X	X	X	X		
		T2	TRIP	SST-2003	I	X	X	X	X	X		
JAXA (FaSTAR Code), Ryoyu Systems	Atsushi Hashimoto	E1	FaSTAR	SA-noft2	II	X	X	X	X	X	X	X
		E2	FaSTAR	SA-noft2	II	X	X	X	X	X	X	X
		E3	FaSTAR	SA-noft2	I	X	X	X	X	X	X	X
		E4	FaSTAR	SA-noft2	I		X	X	X	X	X	
EXA Powerflow	Benedikt König	G1	PowerFLOW	LBM-VLES	Cartesi	o	o	o	o	o	o	o
CFMS, Zenotech, ARA	Andrei Cimpoeu	H2	Edge	EARSM	II	X	X	X	X	X	X	
Mentor Graphics	Chris Watson	I2	FloEFD	k-e Lam-Bremhorst	Cartesian					o	o	o
MetaComp	Uriel Goldberg	J1	CFD++	SA	II	X	X	X	X	X	X	X
		J2	CFD++	SST	II	X	X	X	X	X	X	X
Kawasaki Heavy Industries, Ltd., Aerospace	Taku Nagata	K1	Cflow: 2nd order	SA-noft2 (1st)	II	X	X	X	X	X	X	X
		K2	Cflow: 3rd order	SA-noft2-QCR (1st)	II	X	X	X	X	X	X	X



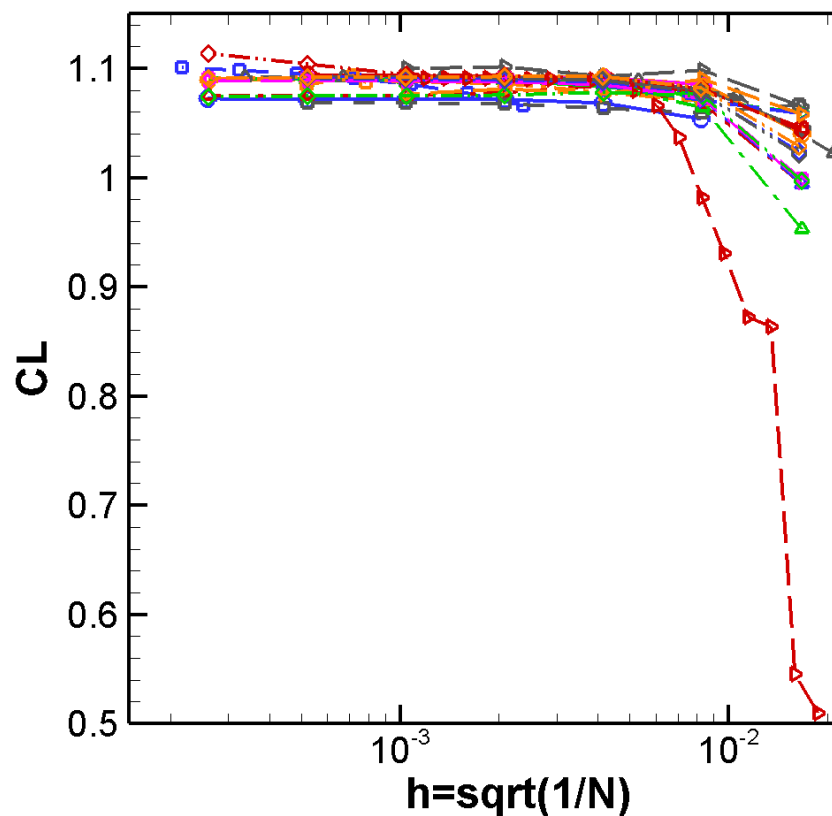
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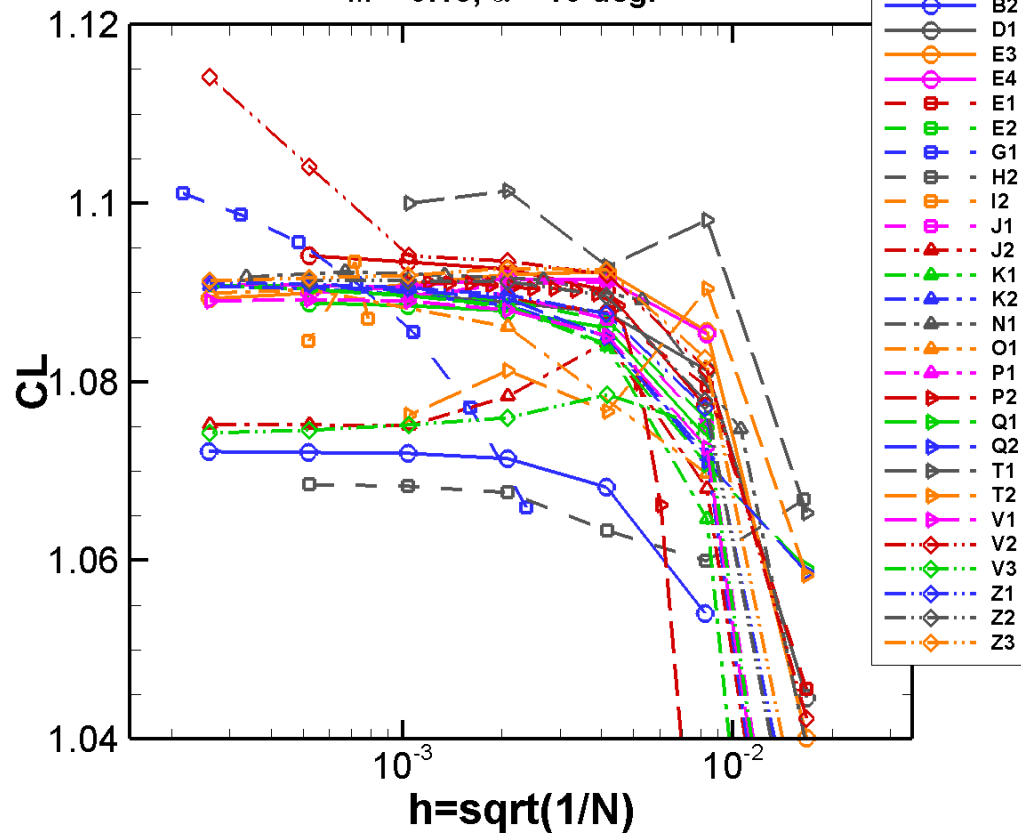
Organisation	Name	Soln ID	Code	Turbulence Model	Case 1							
					Family	113x33	225x65	449x129	897x257	1793x513	3585x1025	7169x2049
CAd Lab, IIS, Bangalore, S&I Engineering Solutions	Balakrishnan	N1	HiFUN	SA (1st)	O-Grid	X	X	X	X	X	X	X
ONERA	David Hue	O1	elsA	SA	II		X	X		X	X	
Boeing, St. Louis	Mori Mani	P1	BCFD	SA	Unstr			X	X	X		
		P2	GGNS	SA-neg	Adapt		Adapted Mesh					
MDOLab, University of Michigan	Joaquim R. R. A. Martins	Q1	SUMad	SA 1stOrder	II	X	X	X	X	X	X	
		Q2	SUMad	SA	II	X	X	X	X	X	X	
Embraer S/A	Rodrigo Felix de Souza	V1	CFD++	SA	II	X	X	X	X	X	X	X
		V2	CMSOft-AERO	SA	II	X	X	X	X	X	X	X
		V3	CFD++	SST	II	X	X	X	X	X	X	X
JAXA, Ryoyu Systems	Yasushi Ito	Z1	TAS	SA-noft2-R-QCR-K1	II	X	X	X	X	X	X	X
		Z2	TAS	SA-noft2-R-QCR-K5	II	X	X	X	X	X	X	
		Z3	TAS	SA-noft2-R-QCR-K10	II	X	X	X	X	X	X	X

## Results:

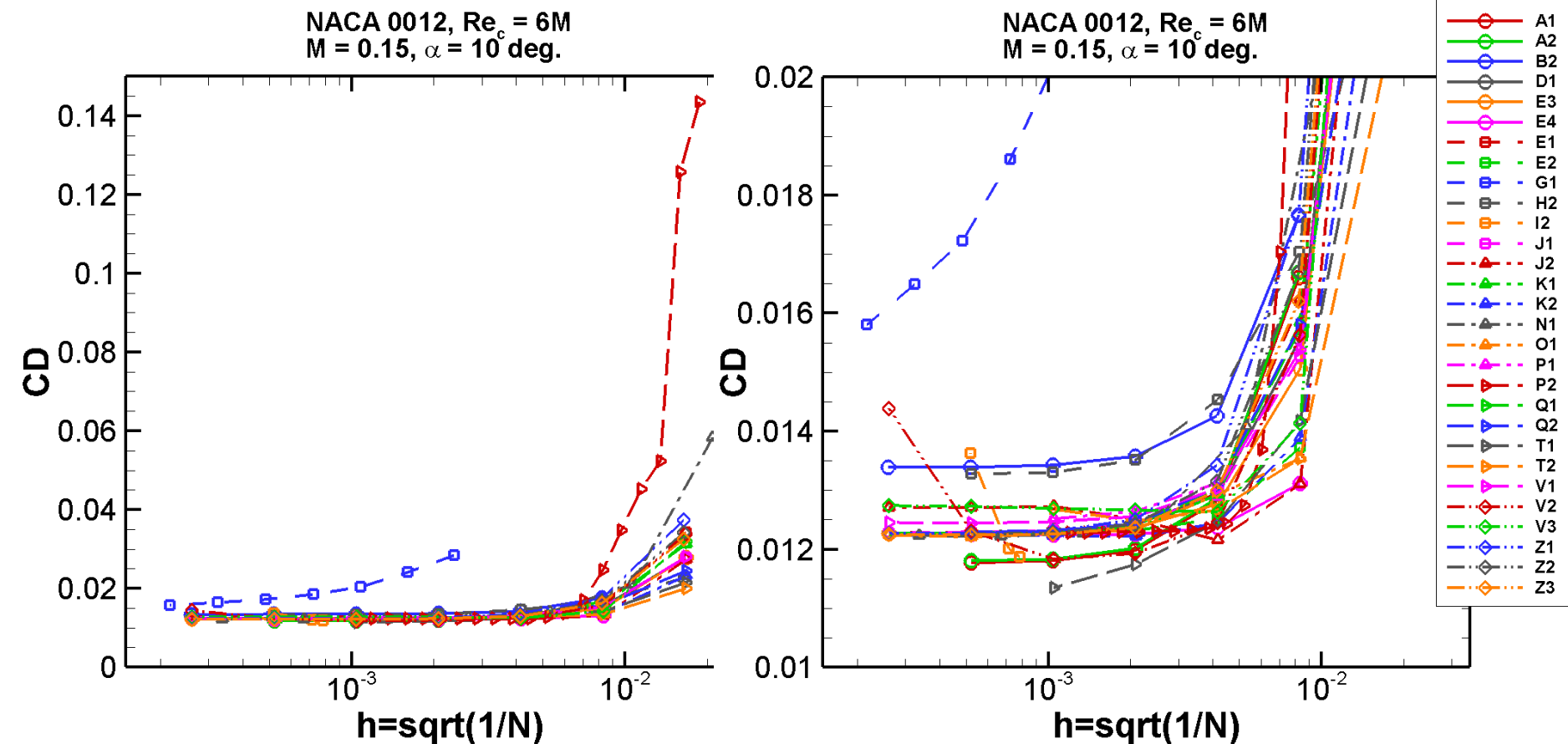
NACA 0012,  $Re_c = 6M$   
 $M = 0.15$ ,  $\alpha = 10^\circ$  deg.



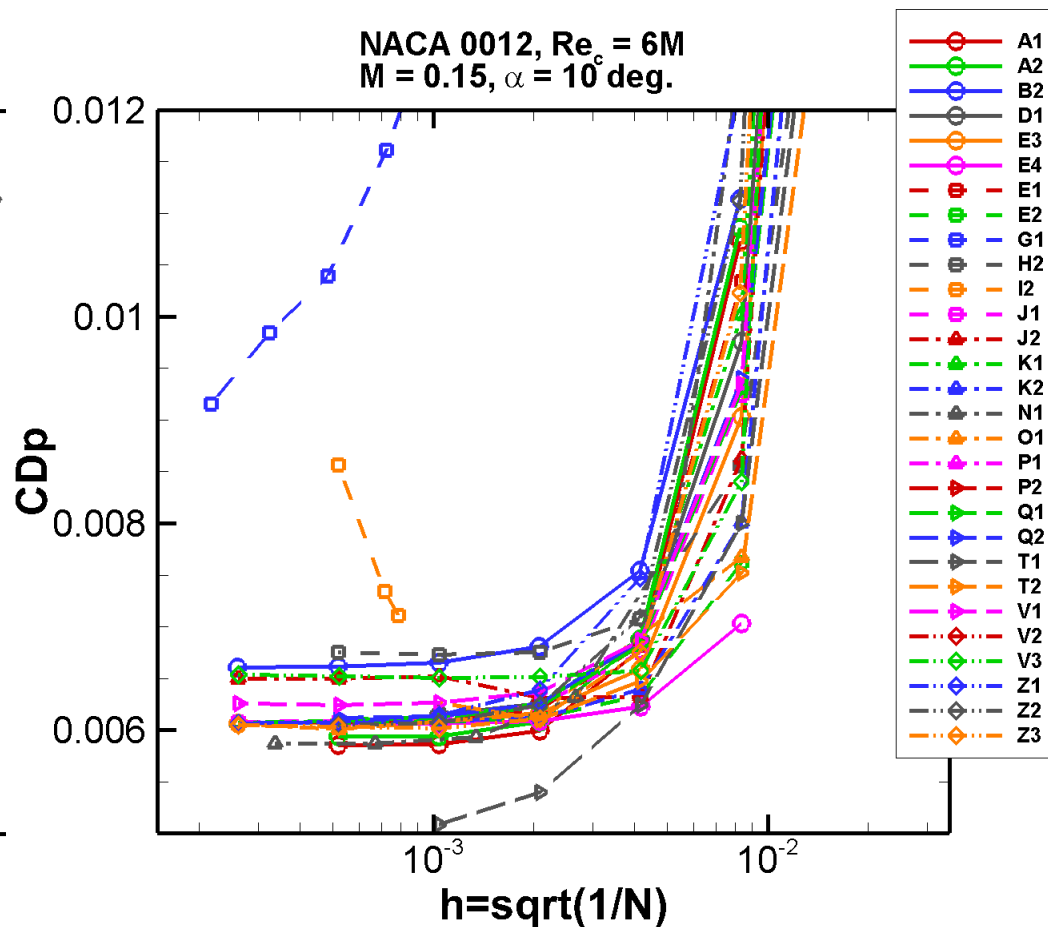
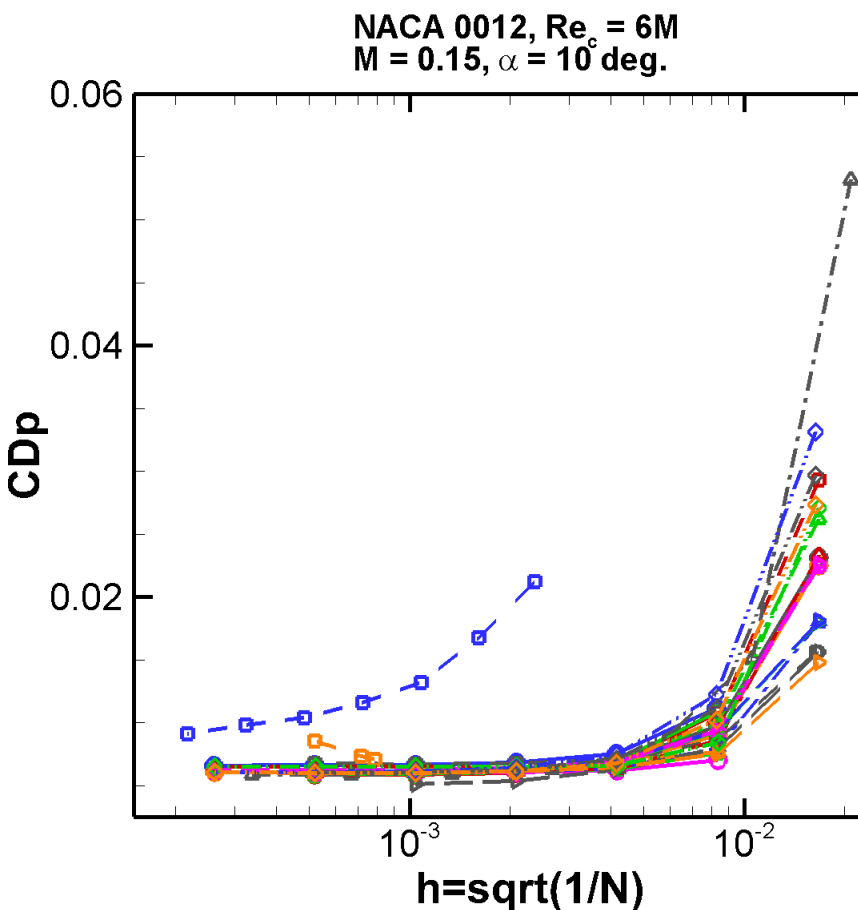
NACA 0012,  $Re_c = 6M$   
 $M = 0.15$ ,  $\alpha = 10^\circ$  deg.



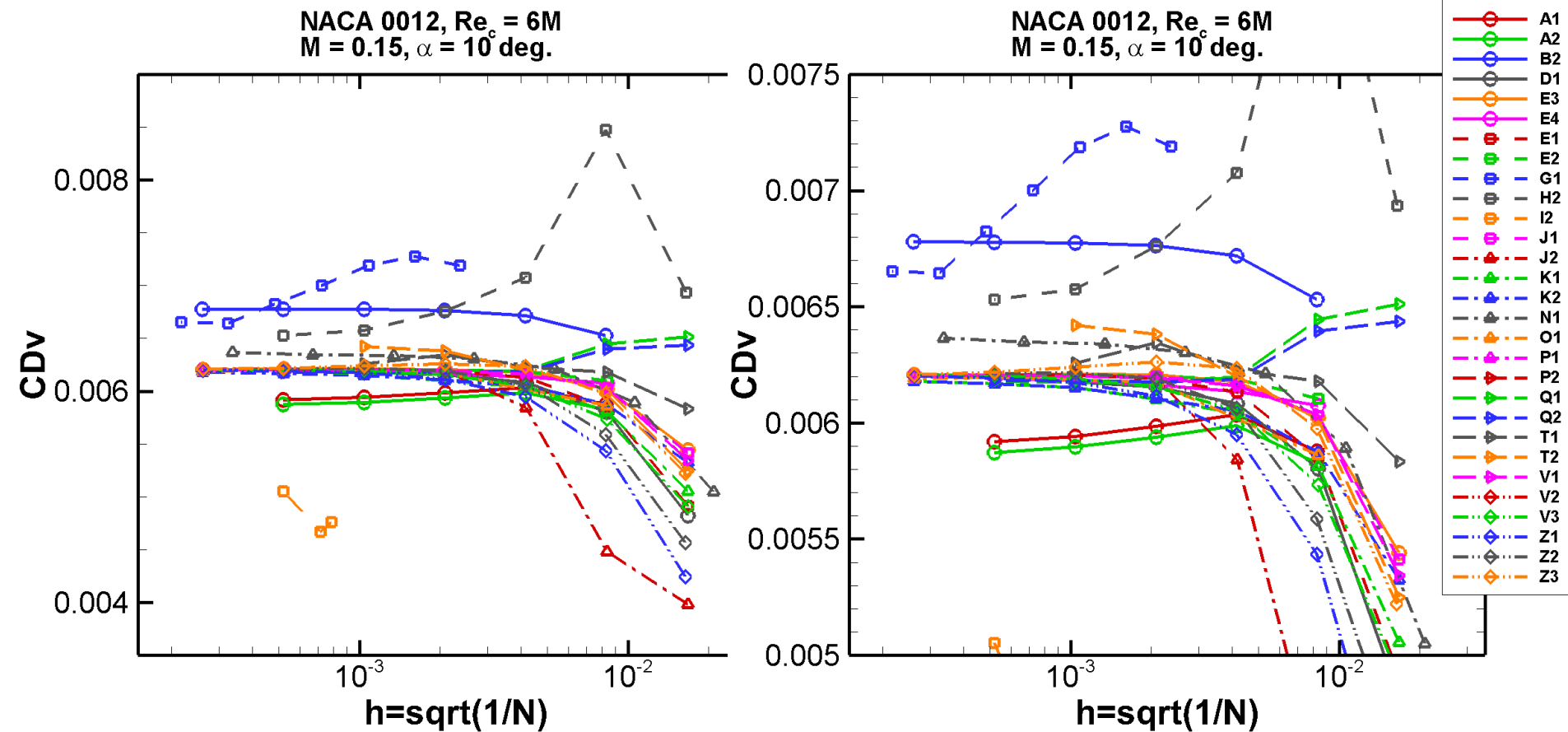
## Results:



## Results:

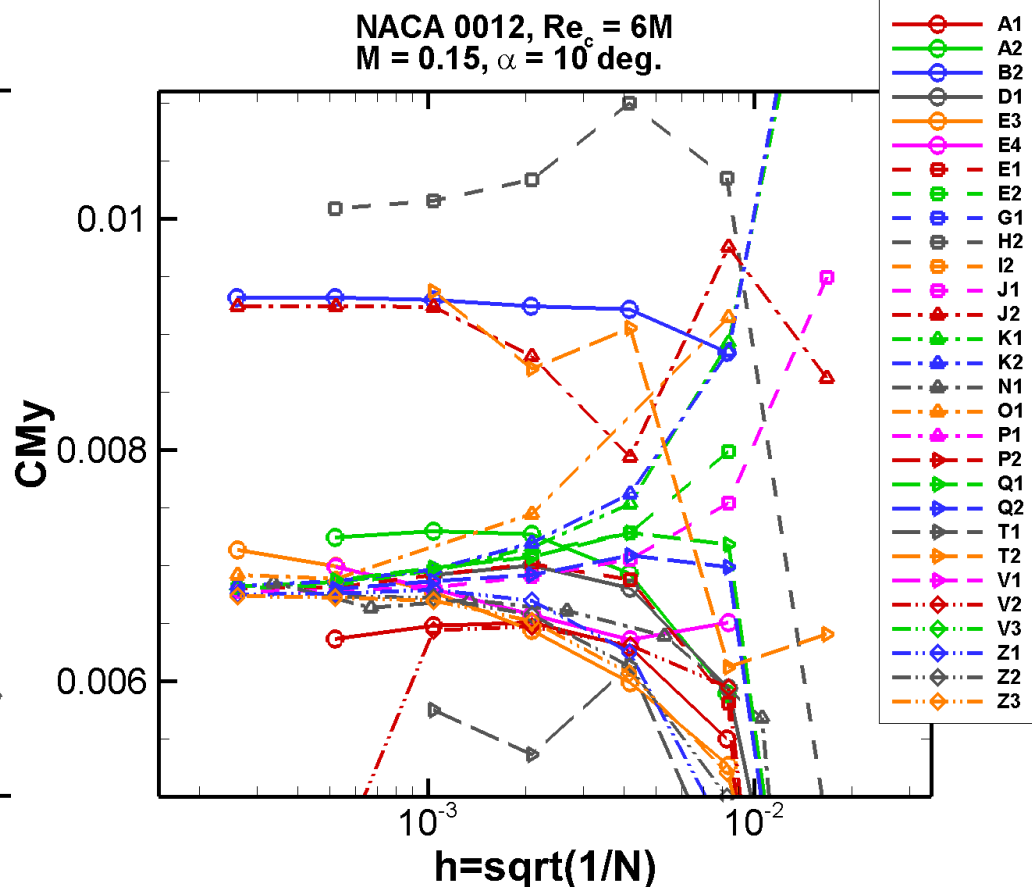
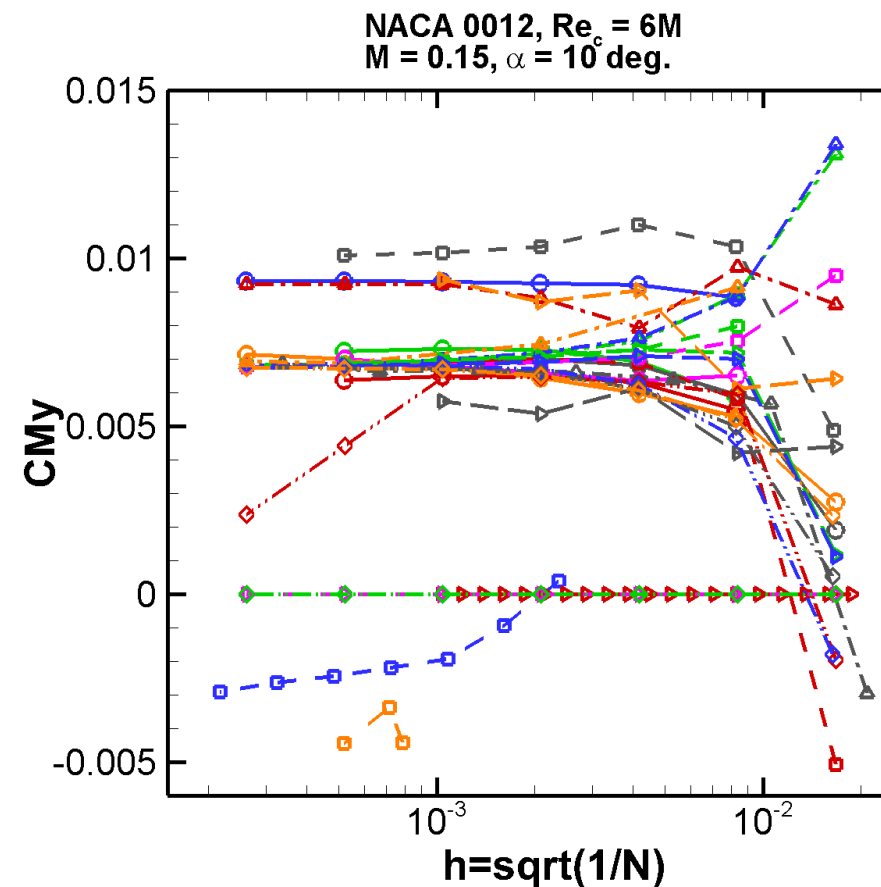


## Results:

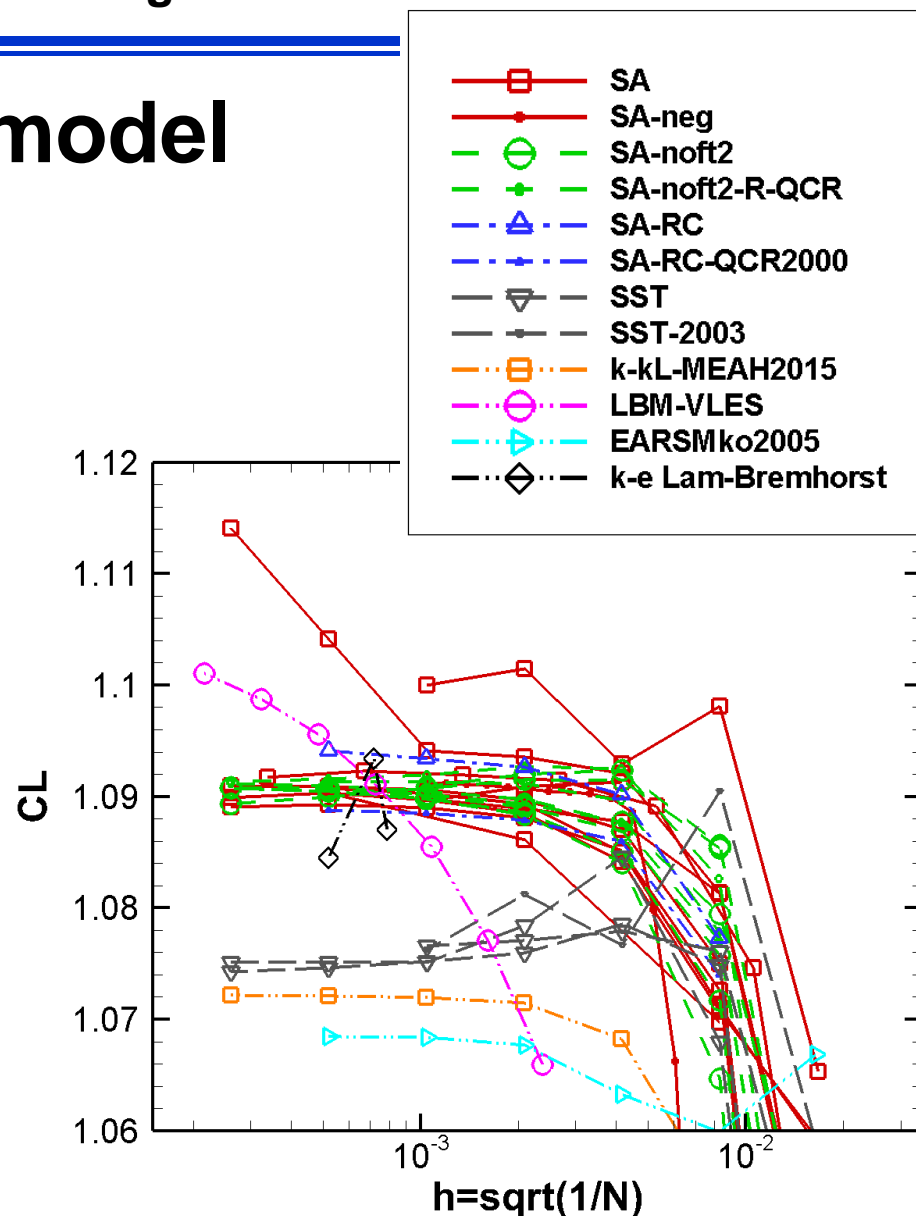
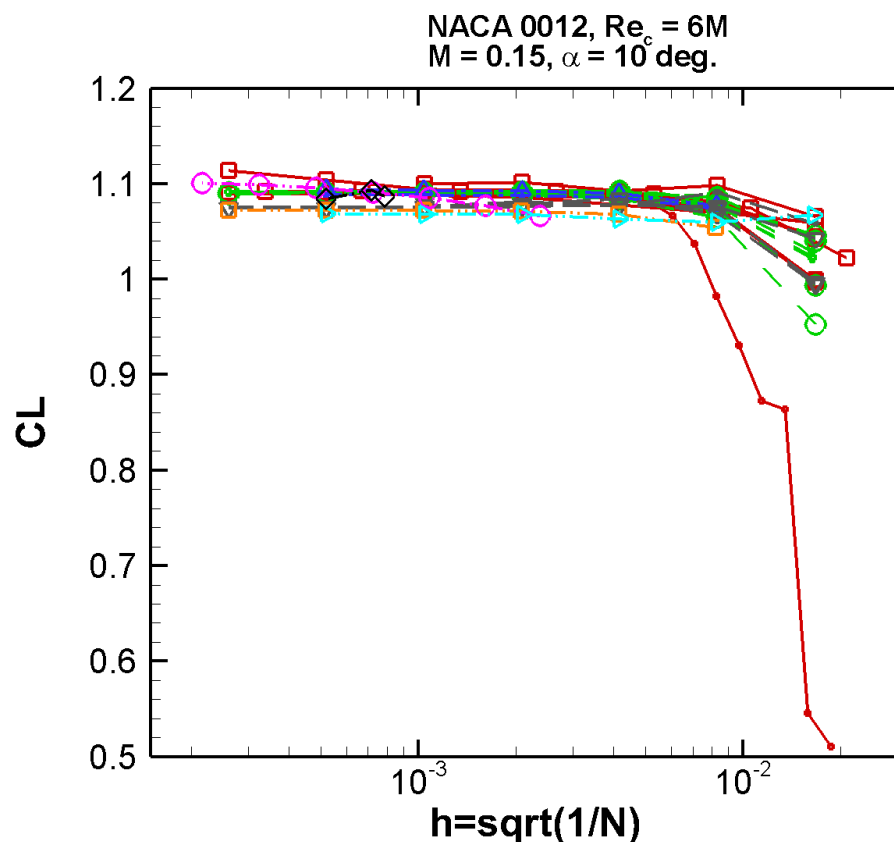




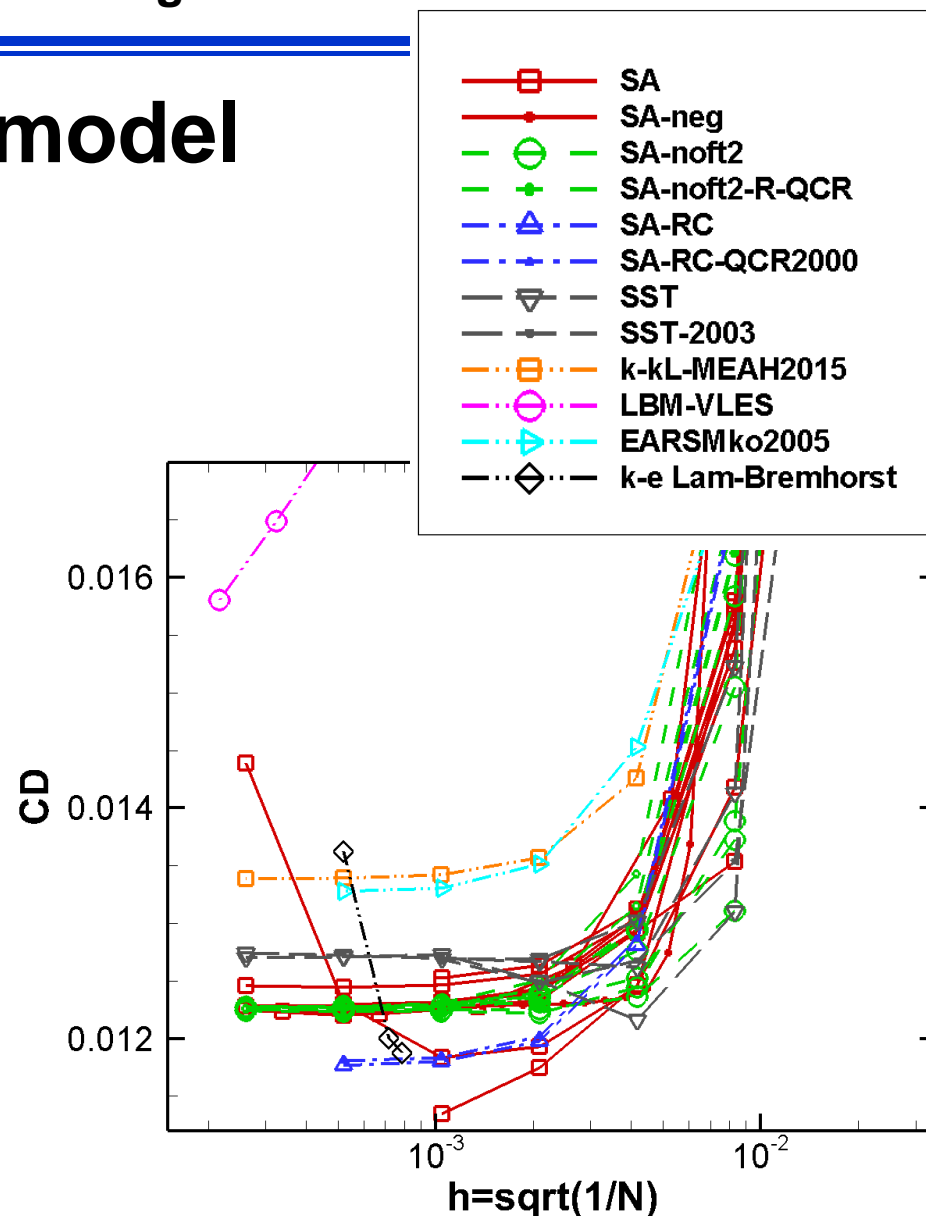
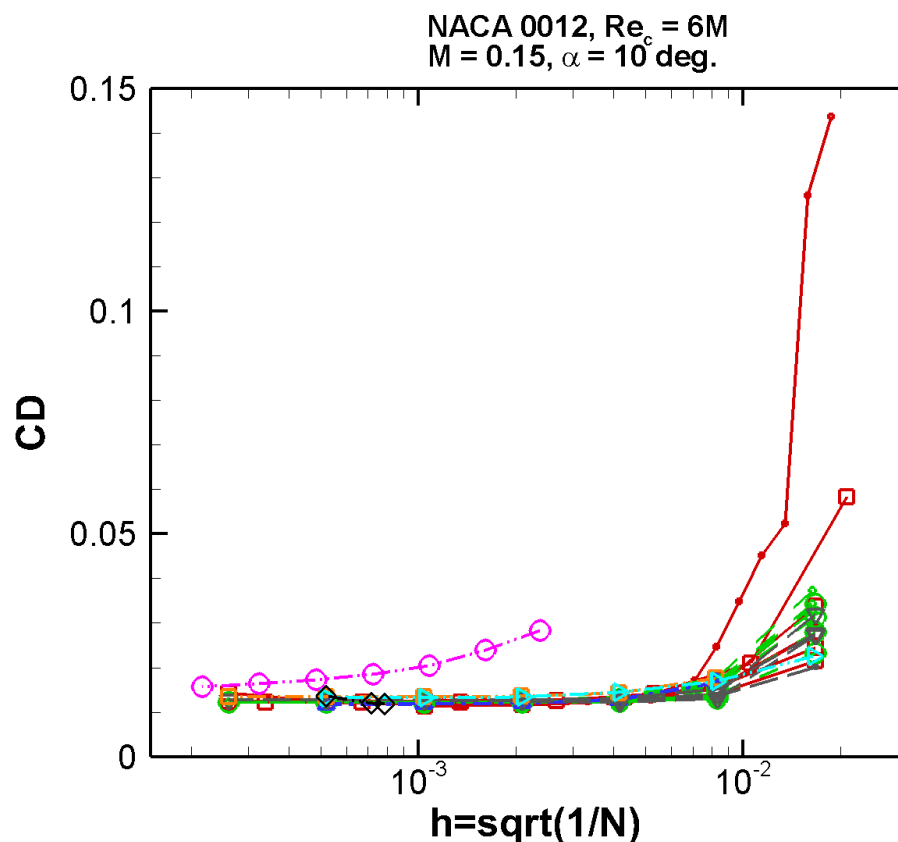
## Results:



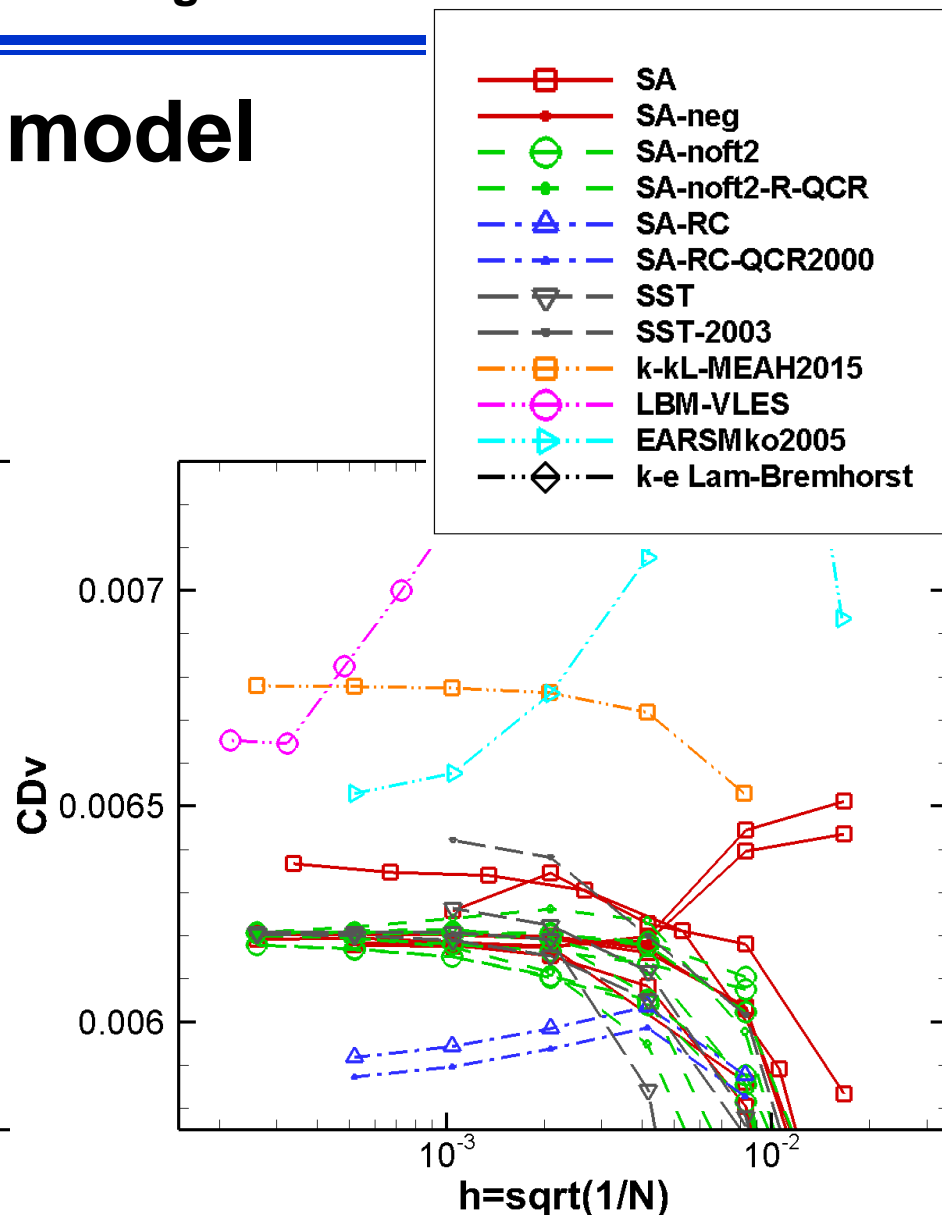
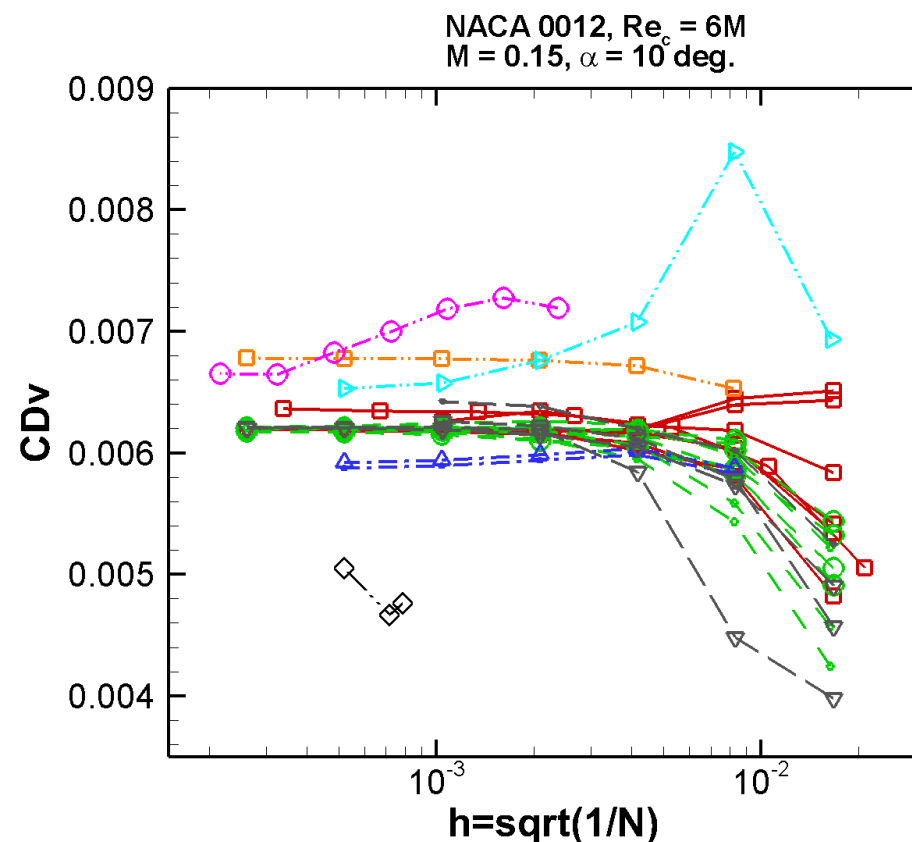
## Effects of turbulence model



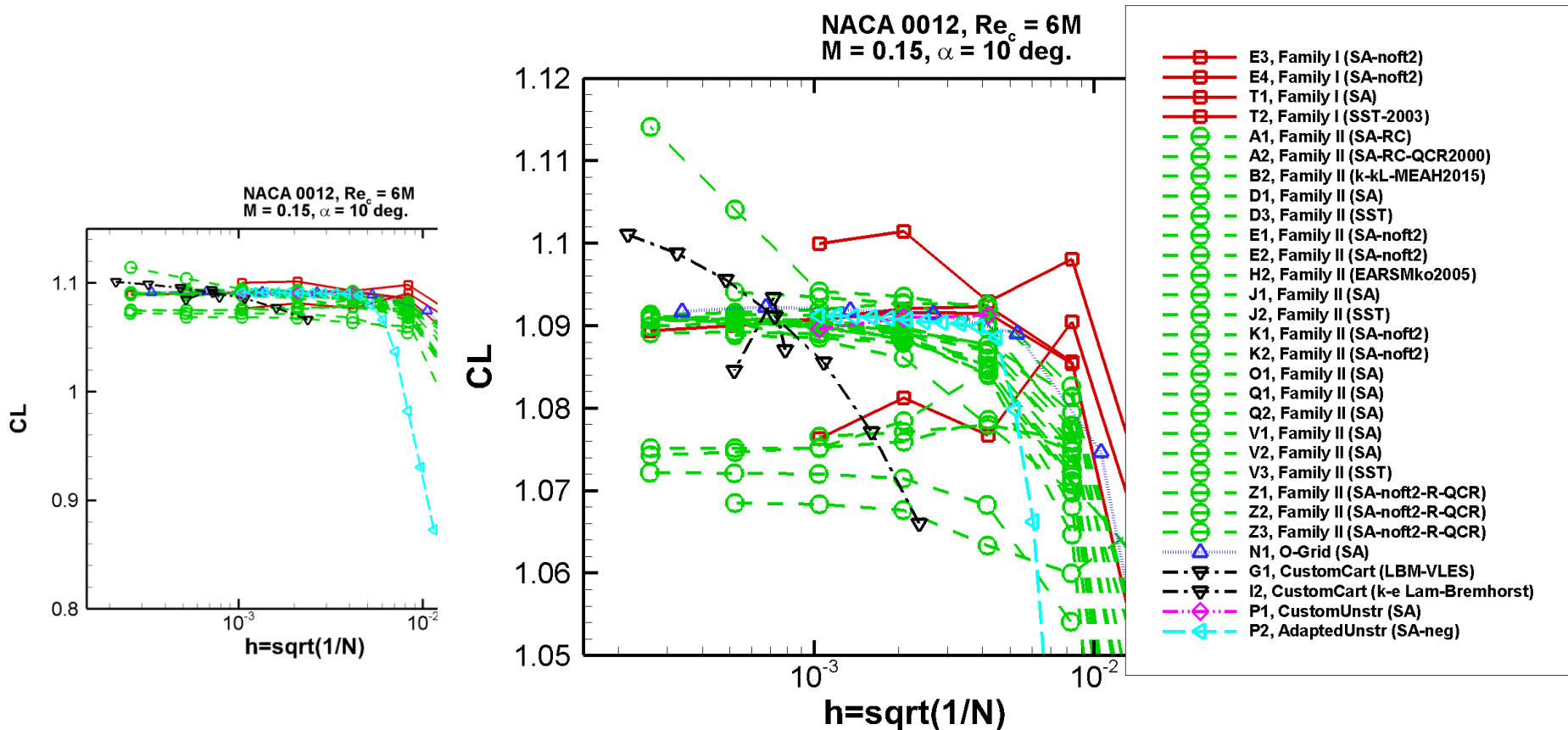
## Effects of turbulence model



## Effects of turbulence model

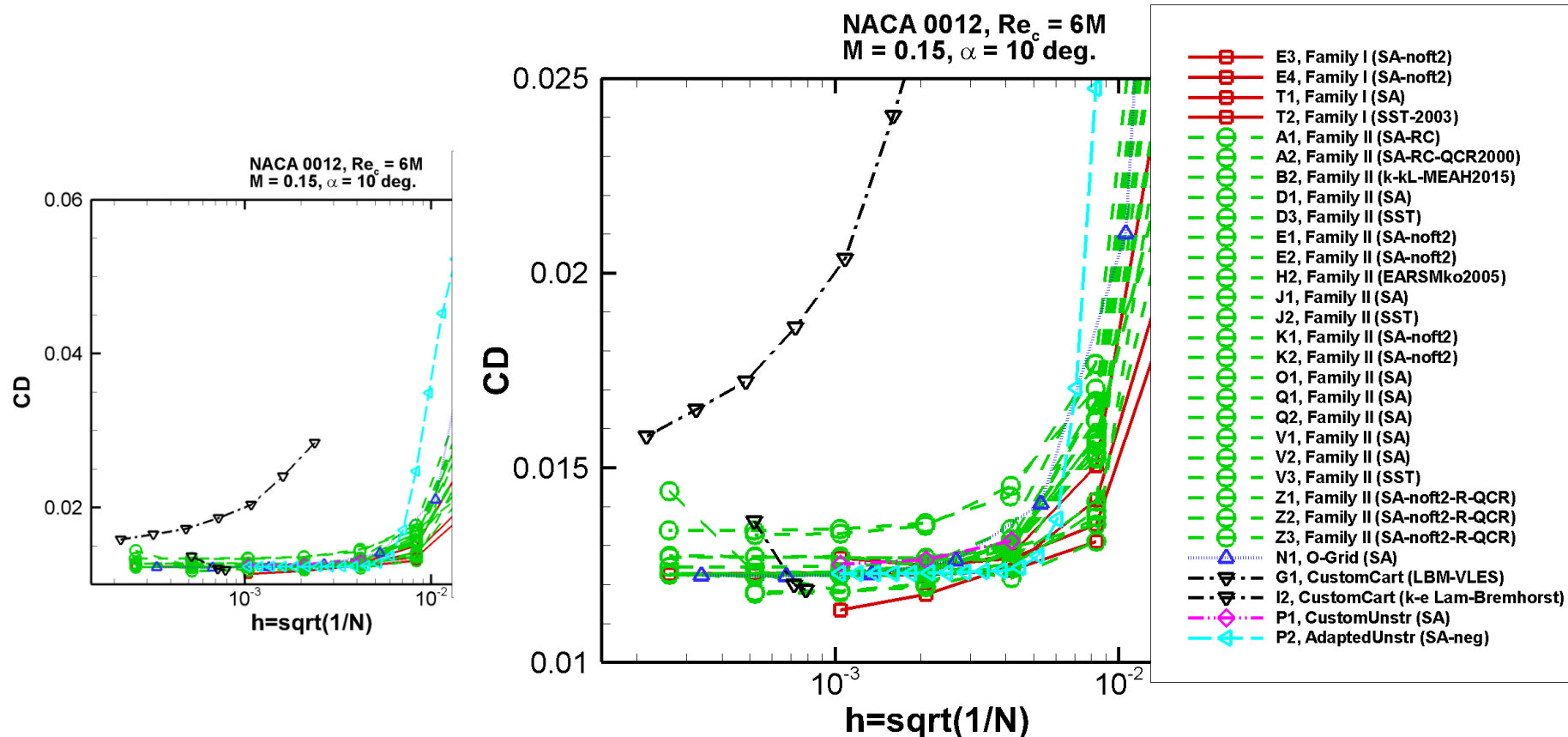


## Effects of grid type

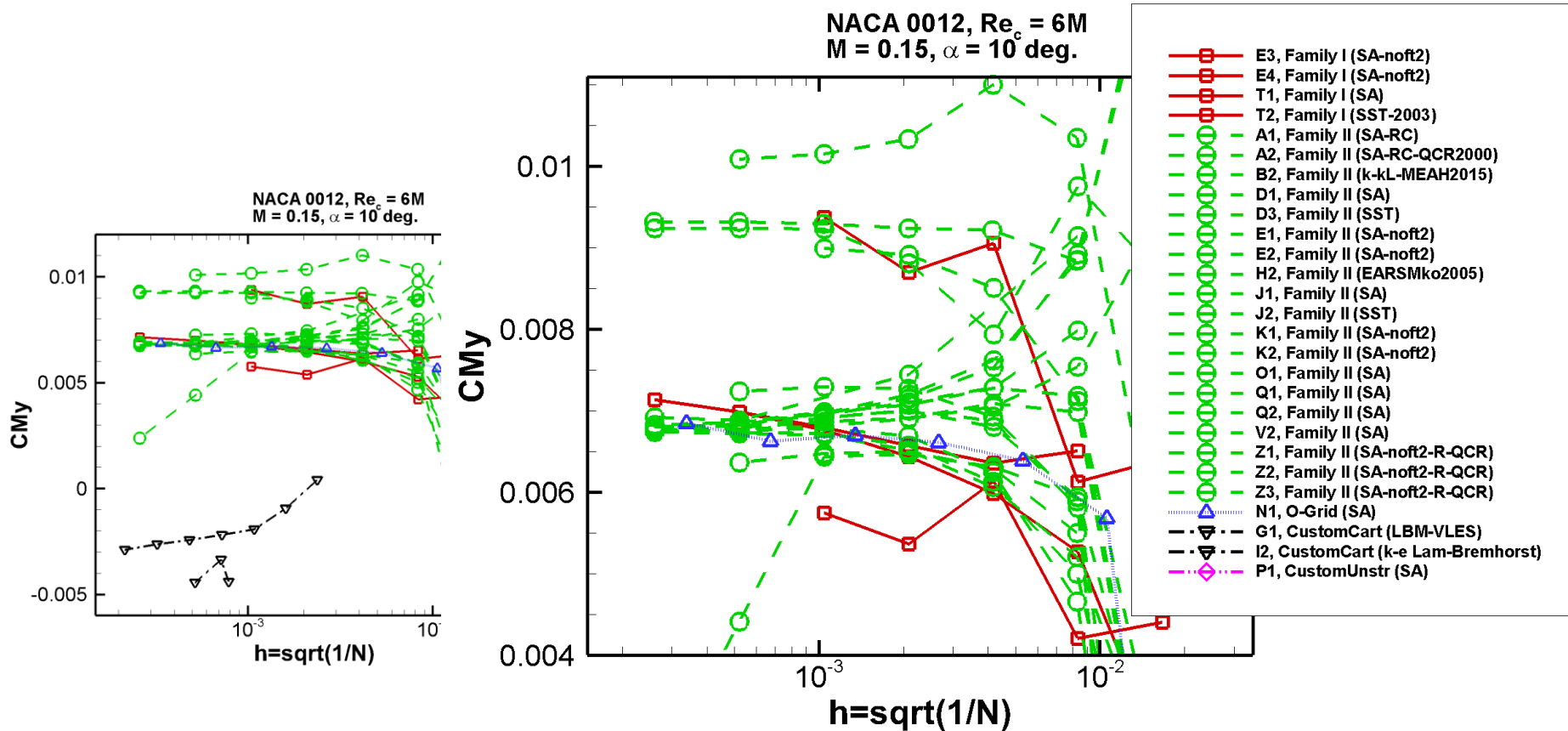




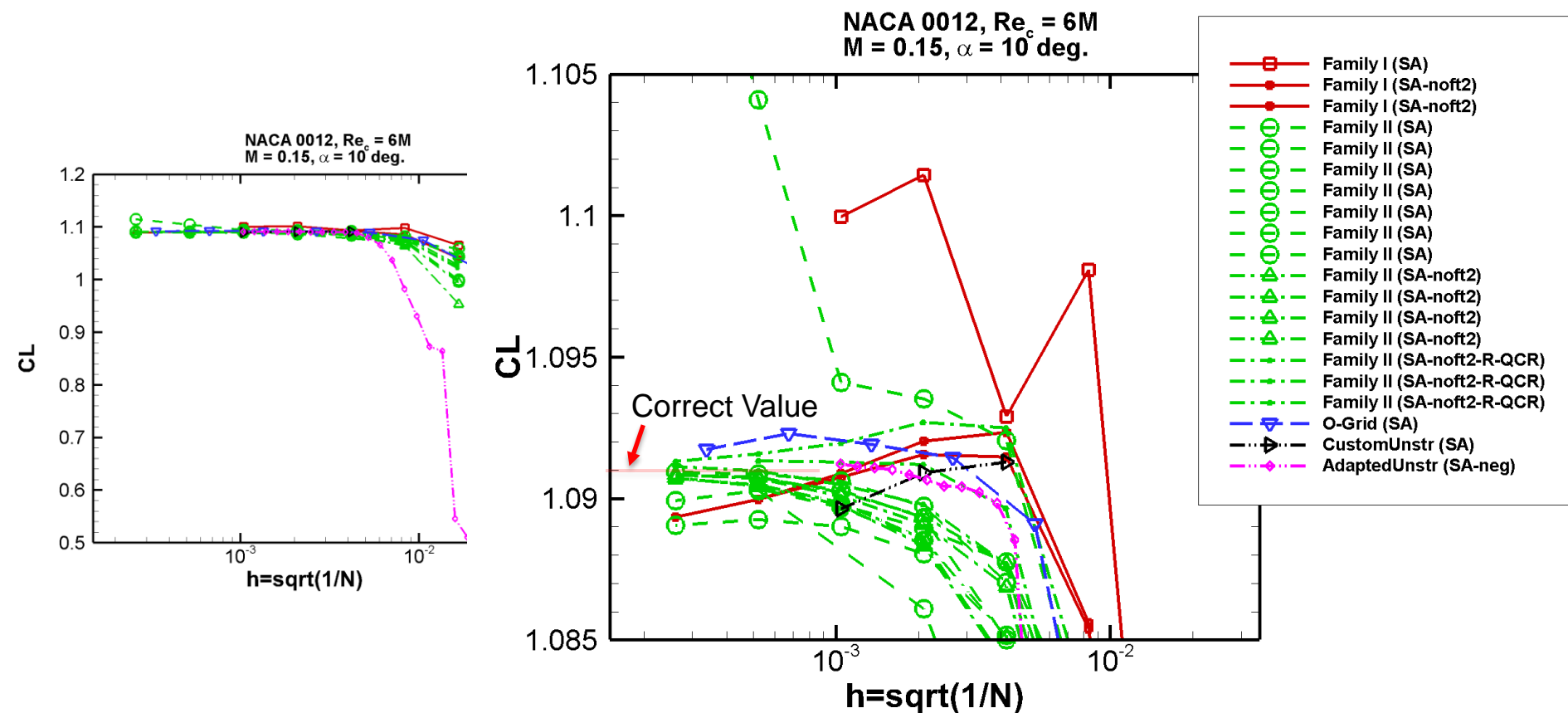
## Effects of grid type



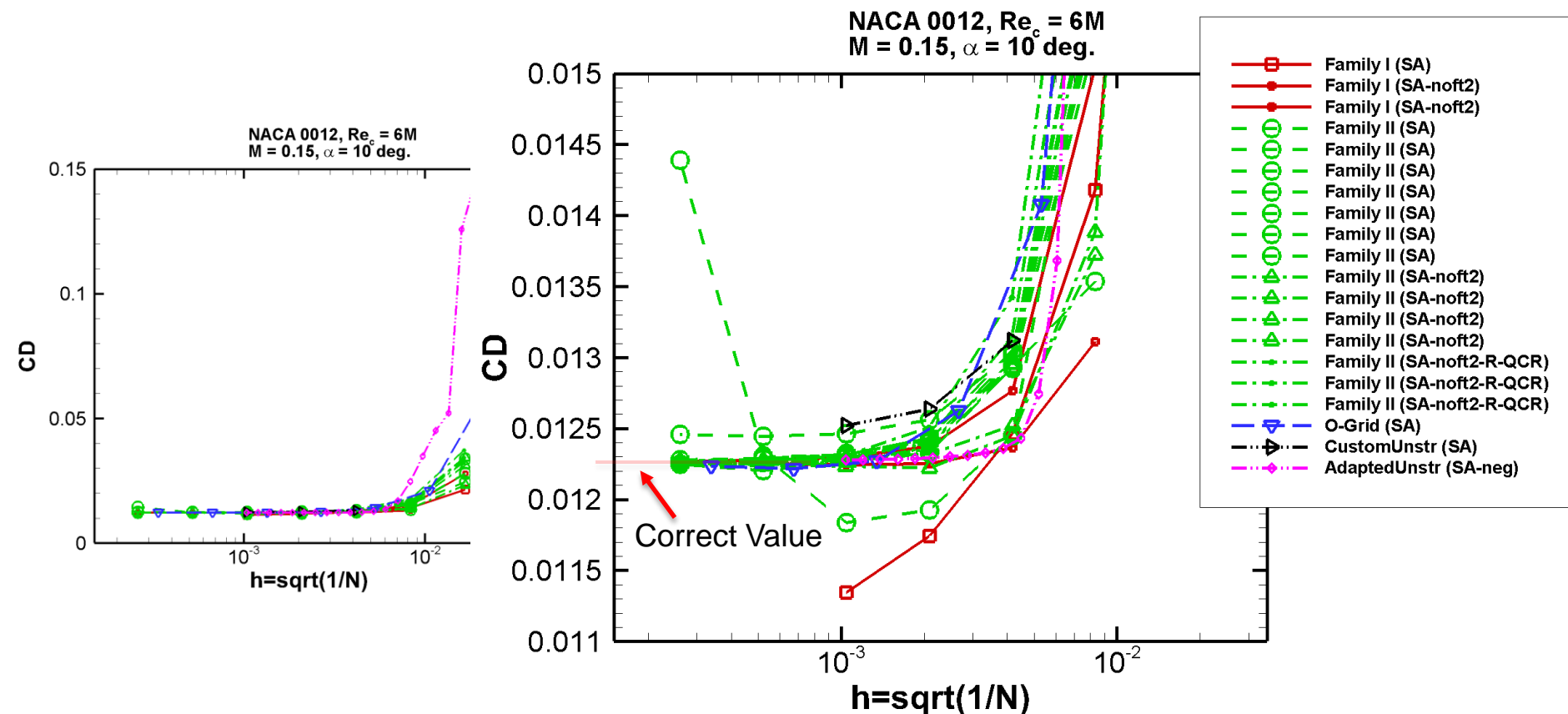
## Effects of grid type



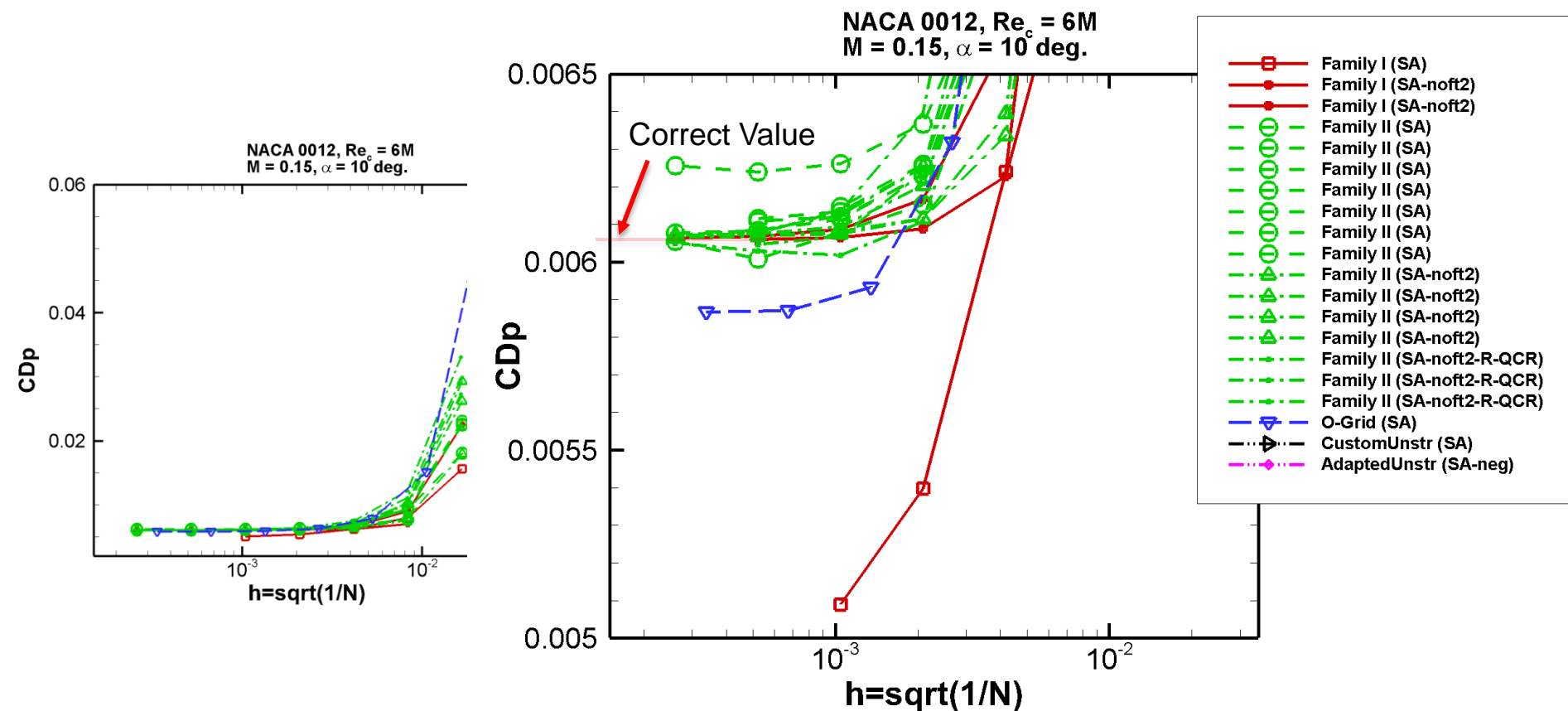
## Effects of grid type: SA only



## Effects of grid type: SA only

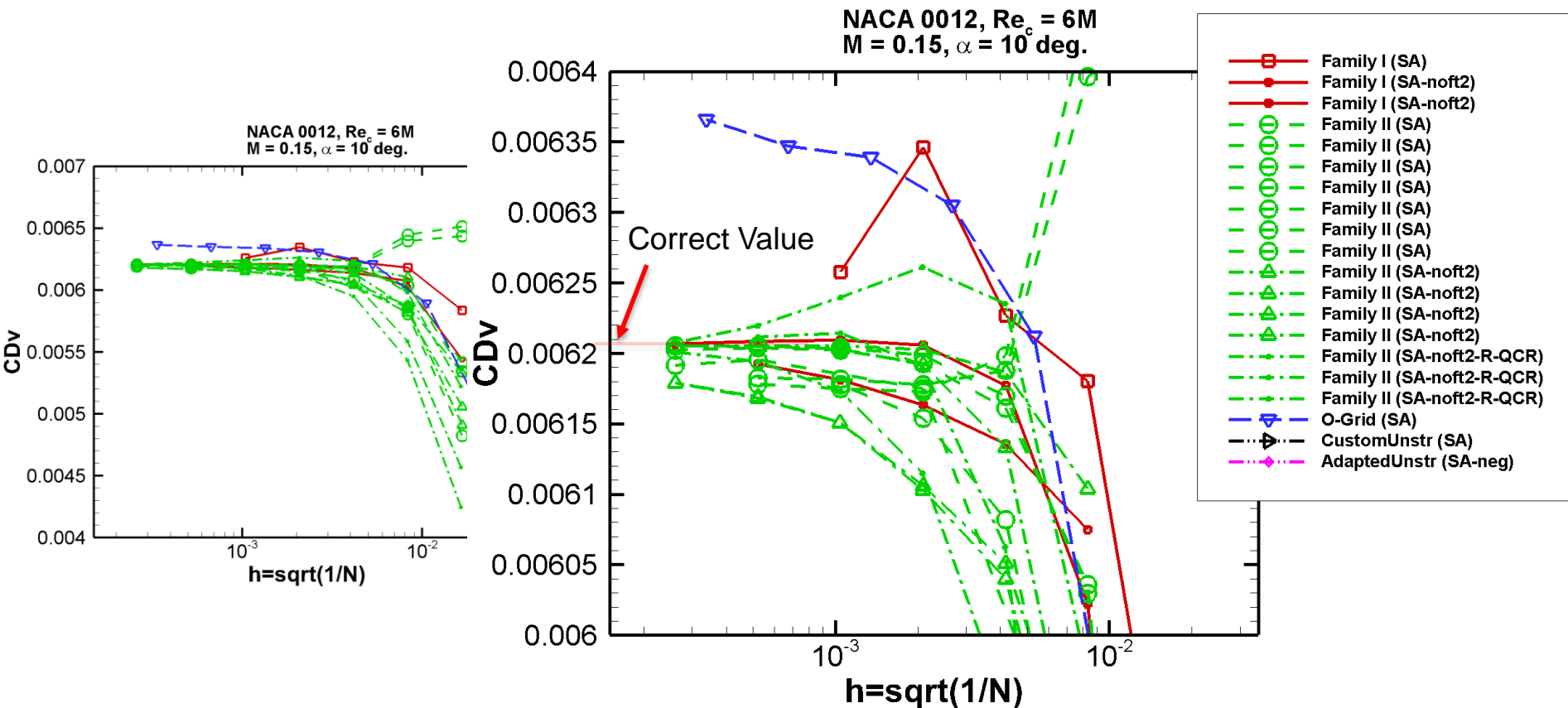


## Effects of grid type: SA only

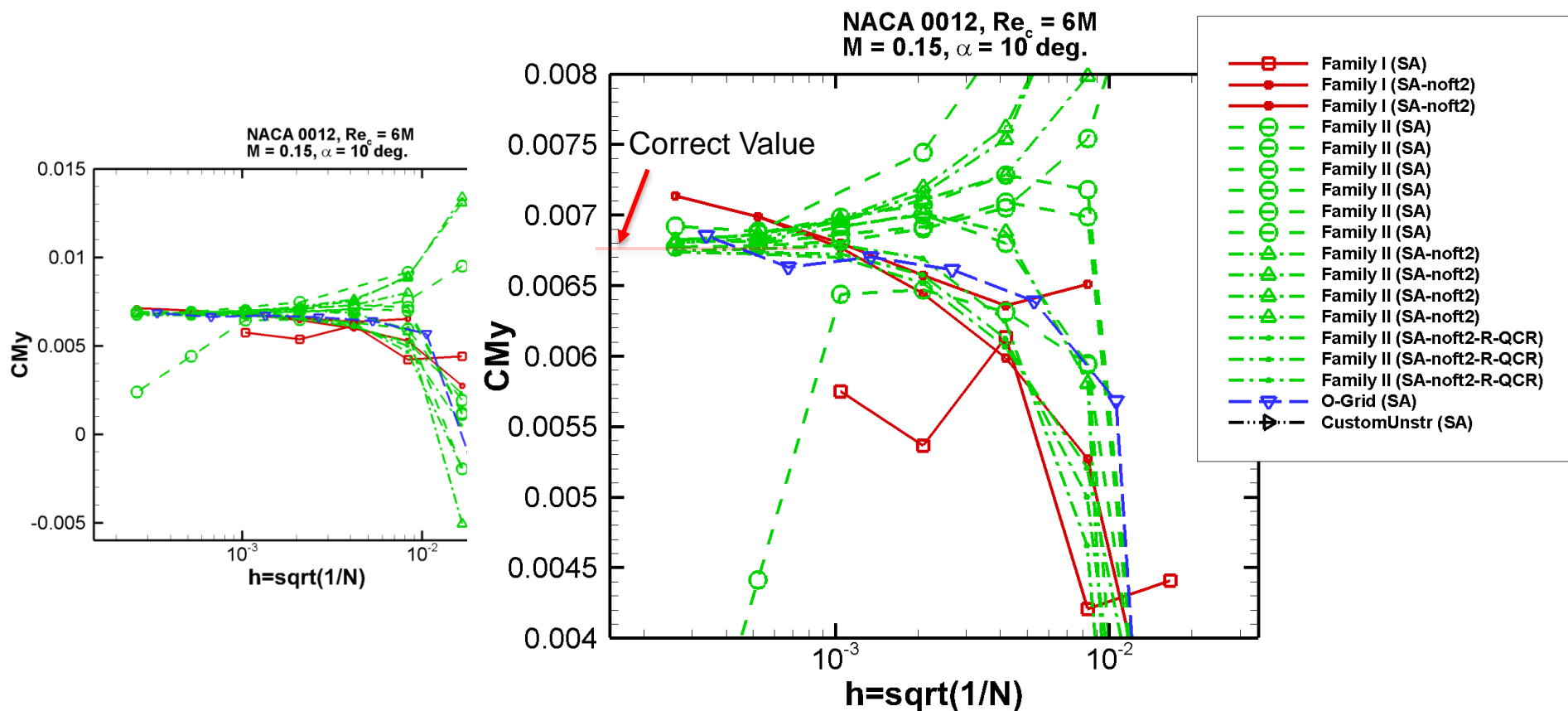




## Effects of grid type: SA only



## Effects of grid type: SA only



## Conclusions for Case #1

- **Grid effects and other numerical issues (iterative convergence?) are still polluting the results**
- **Grid type is important**
  - Cartesian methods appear to either converge slowly or not converge
- **Grid adaptation helps: the single adapted grid case honed in on the (correct) converged values at much lower cell counts**
- **Adequate code verification is still not being done**
  - Ideal approach is to demonstrate order of accuracy using Manufactured Solutions
  - This numerical benchmark allows an easier path to code verification
- **Errors that occur for this simple 2D case are expected to be (much) larger for the more complex 3D cases**
- **Code verification should be a prerequisite for application of a code to analysis, model validation, etc.**

## Thanks go to:

- **Chris Rumsey of NASA Langley for helping with formatting for data files**
- **Ed Tinoco for doing the initial screening of the Case 1 data**
- **All of the DPW-6 participants**



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# Extra Slides



## Effects of order of accuracy of turbulence models

- TBD
  - TBD

**To Be Updated**

## Pressure distributions:

- TBD
  - TBD

**To Be Updated**

## Skin friction distributions:

- TBD
  - TBD

**To Be Updated**



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## Order of accuracy plots for SA models